

# CRACK GROWTH ANALYSIS IN QUASI-BRITTLE MATERIALS USING A NON-LINEAR BOUNDARY ELEMENT FORMULATION

Edson Denner Leonel<sup>1</sup> & Wilson Sergio Venturini<sup>2</sup>

## Abstract

*This paper addresses to analysis of crack propagation in quasi-brittle materials using the boundary element method (BEM). BEM has been widely used to solve many complex engineering problems, especially those where its mesh dimension reduction includes advantages in the modelling. The non-linear formulations developed are based on the dual BEM, in which singular and hyper-singular integral equations are adopted. The first formulation uses the concept of constant operator, in which the corrections on the non-linear system of equations are performed only by applying appropriate tractions along the crack surfaces. The second proposed BEM formulation is an implicit technique based on the use of a tangent operator. This formulation is accurate, stable and always requires less iterations to reach the equilibrium within a given load increment in comparison with the classical approach. Examples of problems of crack growth are shown to illustrate the performance of these two formulations.*

*Keywords: Boundary element method. Non-linear formulation. Tangent operator.*

## 1 INTRODUCTION

During the last few years, fracture mechanics problems have received large interest by the scientific community, because crack growth can explain the collapse of structures. Due to the load values or imposed displacements beyond a critical level, micro-crack concentration increases as well as the structural damage. The micro-cracks growth forms a macro-crack (coalescence phenomenon) who leads to the structural failure. The Hillerborg cohesive crack model is an idealization that represents well this effect, particularly for the analysis of quasi-brittle materials, (HILLERBORG et al., 1976) and (CARPINTERI, 1989). A fictitious crack approximates the dissipation zone since the process begins.

To accurate modelling this effect in complex engineering structures, numerical models are required. The boundary element method (BEM) has already been recognized as an accurate and efficient numerical technique to deal properly with many problems in engineering. Especially, those related to fracture mechanics, where its mesh dimension reduction is an important advantage. Considering these problems, only boundary and crack surface discretizations are required. Consequently, BEM requires less computational effort to generate new elements in order to model crack growth. Some classical BEM formulations were proposed in the literature demonstrating the efficiency and accuracy of this method in fracture problems. In this context, it is worth to mention (CRUSE, 1988) and (CROUCH, 1976), where Green's functions and displacement discontinuity method were adopted, respectively. (PORTELA et al., 1992) developed the dual boundary element method (DBEM), in which singular and hyper-singular integral equations are written for collocation points positioned at the opposite crack surfaces. Considering multiple crack propagation, some BEM models may be found in the literature. The problems of multiple crack-hole interaction, (YAN, 2006) and (KEBIR et al., 2006), and multiple crack interaction, (LEONEL; VENTURINI a, 2010) and (YAN, 2005), for instance, have been successfully analysed. In these applications, however, non-linear crack propagation has not been considered.

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<sup>1</sup> Professor in the Department of Structural Engineering - EESC-USP, edleonel@sc.usp.br

<sup>2</sup> Professor in the Department of Structural Engineering - EESC-USP, *in memoriam*

Non-linear analysis with BEM has been used since the 1970s. The first works were simple but capable to demonstrate that the method can be applied to simulate these kinds of complex problems, (RICARDELLA, 1973) and (KUMAR; MUKHERJEE, 1977). The non-linear solution for the first BEM formulations was based on the application of a correcting stress field, keeping constant all relevant matrices and therefore leading to a large number of iterations. Recently, more reliable solution techniques based on the use of tangent operators have been proposed, (POON et al., 1998), (BOTTA et al., 2005) and (LEONEL; VENTURINI b, 2010). They require less iterations and are more stable and accurate than classical approaches.

Considering non-linear crack growth analysis using BEM, the solution technique adopted for the majority of the researchers is based on iterative schemes that find traction values along crack surfaces that satisfy an adopted criterion, (SALEH; ALIABADI, 1995). This process is simple and all relevant matrices are kept constant during the process. Again, this kind of technique requires a large number of iterations to achieve the equilibrium for a single load increment. Moreover, for the cases of complex pattern of cracks, for instance with a solid containing many micro-crack, this process can be either inaccurate or unstable.

In this work, non-linear crack growth analysis is developed, considering two-dimensional domains and quasi-brittle materials. To deal properly with the non-linear problem, two solution techniques were used. The first approach is the classical procedure, where the corrections on the cohesive crack tractions are performed by applying a non-equilibrated tractions vector. The second approach considered uses a tangent operator, which has been derived, and the system of equations is solved using the Newton–Raphson method. These models are used to analyze cohesive crack propagation problems where the stiffness reduction due the cracks growth is a major problem. Besides comparing the solutions obtained by using the two discussed technique, whenever possible the results are compared with experimental results.

## 2 COHESIVE CRACK MODEL

The linear elastic fracture mechanics has been an important approach to solve many problems in structural engineering, particularly those where the dissipation zone surrounding cracks are reduced enough to be possible to neglect its non-linear effects. However, for quasi-brittle and also for some ductile materials, the damaged zone ahead of the crack tip is large enough to produce non-linear effects that cannot be neglected. The cohesive model is an appropriate approach to take into account these effects. In this model, the dissipation phenomenon is assumed to occur along the crack path ahead of crack tip, therefore reducing by one the dimension of the dissipation zone. The first models where the dissipation zone was reduced either to a line for 2D problems or to a surface for 3D problems are due to (DUGDALE, 1960) and (BARENBLATT, 1962) who have proposed crack models to represent particularly the ductile material behaviour. The cohesive crack model appropriate to quasi-brittle materials is due to (HILLERBORG et al., 1976).

In this work, the dissipation process, which takes place on a region ahead of the crack tip, is approximated by a simple softening law, assumed along a fictitious crack. This law relates the fictitious crack opening displacement,  $\Delta u$ , to tensile surface forces or cohesive forces,  $f_t$ , applied along the crack surfaces. Figure 1 illustrates the cohesive forces distribution for the (HILLERBORG et al., 1976) model. As can be seen in this figure, the cohesive forces acting along the fictitious crack of length equal to  $l_f$  disappear after a critical crack opening value,  $\Delta u_c$ . The crack starts opening when the cohesive forces reaches a critical tensile value  $f_t^c$ . For values of the tensile stresses lesser than  $f_t^c$  the crack remains closed. For values of the crack opening displacement larger than  $\Delta u_c$ , the cohesive forces are zero.

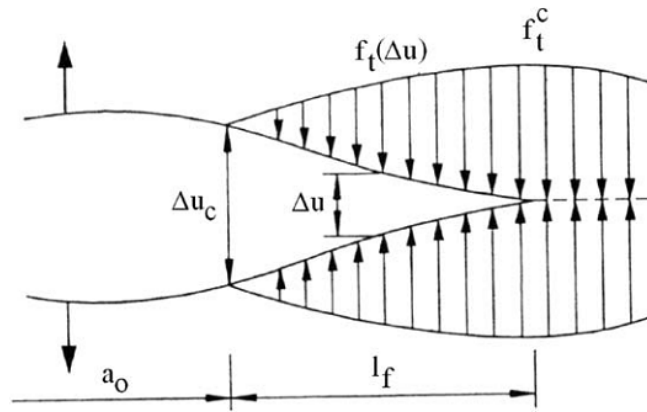


Figure 1 – Cohesive forces distribution. Hillerborg et al. (1976).

Several cohesive crack laws relating cohesive forces and crack opening displacement have already been proposed in the literature. Three of them are often adopted to carry out crack growth analysis in quasi-brittle materials. The simplest law is given by a linear function relating the cohesive forces to the fictitious crack opening displacement smaller than the critical value,  $\Delta u_c$ . For fictitious crack openings larger than  $\Delta u_c$ , cohesive forces are assumed equal to zero, Figure 2a. The relations that represent the linear cohesive law are given by:

$$\begin{aligned} \sigma &= E\varepsilon & \text{if } \varepsilon \leq \varepsilon_c \\ f_t(\Delta u) &= f_t^c \left(1 - \frac{\Delta u}{\Delta u_c}\right) & \text{if } 0 \leq \Delta u \leq \Delta u_c \\ f_t(\Delta u) &= 0 & \text{if } \Delta u > \Delta u_c \end{aligned} \quad (1)$$

An alternative model relating cohesive forces and fictitious crack opening displacement is the bi-linear model, Figure 2b, which is given by the following equations:

$$\begin{aligned} \sigma &= E\varepsilon & \text{if } \varepsilon \leq \varepsilon_c \\ f_t(\Delta u) &= f_t^c - \left(\frac{f_t^c - f_t''}{\Delta u''}\right)\Delta u & \text{if } 0 \leq \Delta u \leq \Delta u'' \\ f_t(\Delta u) &= \frac{f_t''\Delta u}{\Delta u'' - \Delta u_c} + f_t'' \left(1 - \frac{\Delta u''}{\Delta u'' - \Delta u_c}\right) & \text{if } \Delta u'' \leq \Delta u \leq \Delta u_c \\ f_t(\Delta u) &= 0 & \text{if } \Delta u > \Delta u_c \end{aligned} \quad (2)$$

For the bi-linear model, the variables  $f_t''$ ,  $\Delta u''$  and  $\Delta u_c$  are defined by the following expressions:

$$\begin{aligned} f_t'' &= \frac{f_t^c}{3} \\ \Delta u'' &= \frac{0.8G_f}{f_t^c} \\ \Delta u_c &= \frac{3.6G_f}{f_t^c} \end{aligned} \quad (3)$$

The third cohesive crack model considered in this work is represented by an exponential law, Figure 2c. Equation (4) gives the analytical expressions for this cohesive model:

$$\begin{aligned} \sigma &= E\varepsilon && \text{if } \varepsilon \leq \varepsilon_c \\ f_t(\Delta u) &= f_t^c e^{-\frac{f_t^c}{G_f} \Delta u} && \text{if } \Delta u > 0 \end{aligned} \quad (4)$$

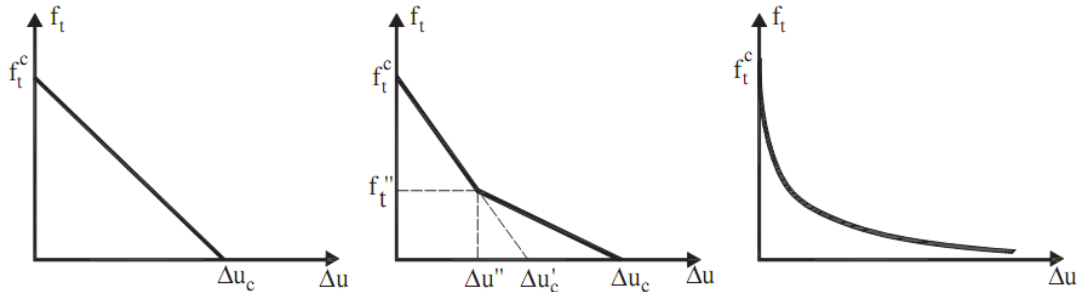


Figure 2 – Cohesive models. (a) linear model; (b) bi-linear model, (c) exponential model.

### 3 DUAL BOUNDARY ELEMENT METHOD (DBEM)

The boundary element method has been widely applied in various engineering fields, such as contact problems, fatigue and fracture mechanics, due to its high precision and robustness in modelling strong stress concentration (i.e. singular stresses and displacements). Considering a two-dimensional homogeneous elastic domain,  $\Omega$ , with boundary,  $\Gamma$ . The equilibrium equation, written in terms of displacements, is given by:

$$u_{i,jj} + \frac{1}{1-2\nu} u_{j,ji} + \frac{b_i}{\mu} = 0 \quad (5)$$

where  $\mu$  is the shear modulus,  $\nu$  is the Poisson's ratio,  $u_i$  are components of the displacement field, and  $b_i$  are body forces. Using Betti's theorem, the singular integral representation, written in terms of displacements can be obtained, with no body forces:

$$c_{lk}(f,c)u_k(f) + \oint_{\Gamma} P_{lk}^*(f,c)u_k(c) d\Gamma = \int_{\Gamma} P_k(c)u_{lk}^*(f,c) d\Gamma \quad (6)$$

where  $P_k$  and  $u_k$  are tractions and displacements on the boundary, respectively,  $\oint$  indicates the Cauchy principal value, the free term  $c_{lk}$  is equal to  $\delta_{lk}$  for smooth boundaries, and  $P_{lk}^*$  and  $u_{lk}^*$  are the fundamental solutions for tractions and displacements.

Equation (6) is sufficient to construct the system of algebraic equations to analyse 2D elastic domains. For solids with cracks, however, using only this equation to assemble the system of equations will lead to a singular matrix as both crack surfaces are located along the same geometrical path. Although possible using only the singular integral representation, Eq. (6) requires the definition of a finite gap between the two crack surfaces and a very accurate integral scheme to compute the integral along the quasi singular elements.

Several BEM formulations have been proposed in the literature to properly deal with crack problems, as discussed in the introduction part. The DBEM formulation is probably the most popular BEM formulation to analysis of arbitrary crack growth. In this formulation, singular integral

representation, Eq. (6), is adopted to determine the algebraic representation related to the collocation points defined along one crack surface, while hyper-singular integral representation is chosen to obtain the algebraic representation related to the collocation points placed along the opposite crack surface. Singular representation is applied to the collocation points on the boundary, which is sufficient to obtain the required algebraic relations.

The hyper-singular integral, written in terms of traction, is obtained from Eq. (6). First this equation, written for an internal collocation point, is differentiated to obtain the integral representation in terms of strains. Then, using the Hooke's law, the stress integral representation is achieved. Finally, the integral representation of stresses for a boundary collocation point is obtained by carrying out the relevant limits. The Cauchy formula is applied to obtain the traction representation as follows:

$$\frac{1}{2} P_j(f) + \eta_k \int_{\Gamma} S_{kij}(f, c) u_k(c) d\Gamma = \eta_k \int_{\Gamma} D_{kij}(f, c) P_k(c) d\Gamma \quad (7)$$

where  $\int_{\Gamma}$  indicates the Hadamard finite part, and terms  $S_{kij}$  and  $D_{kij}$  contain the new kernels computed from  $P_{lk}^*$  and  $u_{lk}^*$ .

Equations (6) and (7) are, as usual, transformed to algebraic relations by dividing the boundary and the crack surfaces into elements along which displacements and tractions are approximated. Besides that, one has to select a convenient number of collocation points to obtain the algebraic representations. The algebraic equations for boundary nodes are calculated using boundary collocation points either at the element ends, therefore coincident with nodes, or along the element when displacement and traction discontinuities are to be enforced. Thus, using the discretized form of Eq. (6), applied only to boundary collocation points, the usual system of algebraic equations can be obtained, relating boundary values, as follows:

$$H_b^b U_b + H_b^f U_f = G_b^b P_b + G_b^f P_f \quad (8)$$

where  $U_b$  and  $U_f$  are displacements assigned to boundary (b) and to crack surface nodes (f),  $P_b$  gives the boundary tractions, while  $P_f$  represents the tractions acting along the crack surfaces;  $H_b^b, H_b^f, G_b^b$  and  $G_b^f$  are the corresponding matrices to take into account displacements and tractions effects, the subscript b indicates that the collocation point is on the boundary and the superscripts specify the boundary (b) or crack surface (f) values.

On the crack surfaces, boundary elements, nodes and collocation points at each crack lip have to be defined. Thus, for the crack surfaces one has two opposite collocations points that originate four algebraic independent relations, corresponding to four unknown crack surface values, two displacements and two tractions. Moreover, as the hyper-singular representation Eq. (7) is considered, it is convenient to use collocation points defined along the element and not coincident with the discretization nodes. The node values of crack displacements and tractions are kept at the element end. Thus, from the discretized forms of Equations (6) and (7), the set of algebraic equations below can be written:

$$H_f^b U_b + H_f^f U_f = G_f^b P_b + G_f^f P_f \quad (9)$$

where the subscript  $f$  in the matrices  $H_f^x$  and  $G_f^x$  indicates equation written for collocation points along the crack surface.

The DBEM formulation adopted in this paper uses continuous and discontinuous linear elements along the external boundary and only discontinuous linear elements along the crack surfaces. The integrals in Eq. (6) are evaluated by using a Gauss–Legendre numerical scheme accomplished with a sub-element technique. The integrals appearing in Eq. (7) are calculated using analytical

expressions. Based on these procedures, Equations (6) and (7) are transformed to algebraic representations with very low integration error.

Using algebraic equations (8) and (9) together with the cohesive crack model described in the previous section one can develop an appropriate algorithm to analyze crack growth problems as will be shown in the next section.

#### 4 SOLUTION TECHNIQUE. TANGENT OPERATOR

The analysis of quasi-brittle fractured solids leads to the solution of a non-linear problem that always requires the use of an iterative process within each time interval or load step. In the context of BEM, a technique based on the use of constant matrices is often adopted, (SALEH; ALIABADI, 1995), with satisfactory results. At each load increment the algebraic equations are kept constant. The only modification required is to reapply a non-equilibrated traction vector along the crack surfaces to reestablish the equilibrium lost when the constitutive model is imposed. This technique is simple, but usually requires a large number of iterations to achieve the equilibrium within a load increment, which might lead to unstable solutions. Non-linear BEM formulations based on more elaborated solution techniques have shown to be more accurate and stable. For instance, non-linear BEM formulations based on the use of tangent operators have demonstrated to lead to accurate and stable solutions when modelling complex plastic and damaged solids characterized by exhibiting localization and bifurcation phenomena, (BOTTA et al., 2005) and (LEONEL; VENTURINI b, 2010).

To derive the formulation based on the use of tangent operator, the equations (8) and (9) have to be modified. The tractions and displacements values on the crack surfaces will be separated according its position, at the right or left crack surface.

$$H_b^b U_b + H_b^{fr} U_r^f + H_b^{f1} U_1^f = G_b^b P_b + G_b^{fr} P_r^f + G_b^{f1} P_1^f \quad (10)$$

$$H_f^b U_b + H_f^{fr} U_r^f + H_f^{f1} U_1^f = G_f^b P_b + G_f^{fr} P_r^f + G_f^{f1} P_1^f \quad (11)$$

where the subscripts  $r$  and  $l$  are related to right and left crack surfaces, respectively.

Equations (10) and (11) can be further modified. First by transforming the crack surface values, displacements and tractions, into local coordinates  $(n, s)$ , in which  $n$  and  $s$  are coordinate axes perpendicular and parallel to each crack surface, respectively:

$$H_b^b U_b + H_b^{frs} U_{rs}^f + H_b^{frm} U_m^f + H_b^{f1s} U_{1s}^f + H_b^{f1n} U_{1n}^f = G_b^b P_b + G_b^{frs} P_{rs}^f + G_b^{frm} P_m^f + G_b^{f1s} P_{1s}^f + G_b^{f1n} P_{1n}^f \quad (12)$$

$$H_f^b U_b + H_f^{frs} U_{rs}^f + H_f^{frm} U_m^f + H_f^{f1s} U_{1s}^f + H_f^{f1n} U_{1n}^f = G_f^b P_b + G_f^{frs} P_{rs}^f + G_f^{frm} P_m^f + G_f^{f1s} P_{1s}^f + G_f^{f1n} P_{1n}^f \quad (13)$$

After these modifications, the relative displacements on the crack surfaces can be included in the formulation. The crack relative displacements on directions parallel and perpendicular to the crack surfaces,  $u_s$  and  $u_n$  respectively, are used to replace the displacement components along the left crack side.

$$u_s = U_{1s}^f + U_{rs}^f \quad \rightarrow \quad U_{1s}^f = u_s - U_{rs}^f \quad (14)$$

$$u_n = U_{1n}^f + U_{rn}^f \quad \rightarrow \quad U_{1n}^f = u_n - U_{rn}^f \quad (15)$$

After these modifications, equations (12) and (13) can be rewritten as:

$$\begin{aligned}
& H_b^b U_b + (H_b^{frs} - H_b^{f1s}) U_{rs}^f + (H_b^{frm} - H_b^{f1n}) U_{rn}^f + H_b^{f1s} u_s + H_b^{f1n} u_n = \\
& = G_b^b P_b + G_b^{frs} P_{rs}^f + G_b^{frm} P_{rn}^f + G_b^{f1s} P_{1s}^f + G_b^{f1n} P_{1n}^f
\end{aligned} \tag{16}$$

$$\begin{aligned}
& H_f^b U_b + (H_f^{frs} - H_f^{f1s}) U_{rs}^f + (H_f^{frm} - H_f^{f1n}) U_{rn}^f + H_f^{f1s} u_s + H_f^{f1n} u_n = \\
& = G_f^b P_b + G_f^{frs} P_{rs}^f + G_f^{frm} P_{rn}^f + G_f^{f1s} P_{1s}^f + G_f^{f1n} P_{1n}^f
\end{aligned} \tag{17}$$

Equations (16) and (17) represent the equilibrium of a solid containing cracks. For the case of non-linear fracture problems, these two last equations have to be written and solved within the context of incremental problems. Thus, Equations (16) and (17) have to be conveniently rewritten in its incremental forms. Firstly, these equations have to be written in rates and then transformed into their incremental forms by performing the relevant time integrals over a typical time increment  $\Delta t = \Delta t_{n+1} - \Delta t_n$ . For the postulated problem, the corresponding incremental forms of Equations (16) and (17) is simple and obtained by replacing all boundary and crack values  $x$  by their increments  $\Delta x$ .

For a given load increment, Equations (16) and (17) can be further modified by applying the known boundary conditions. As usual in BEM formulations, all unknown boundary values are stored in a vector  $\Delta x$  and cumulate the known boundary values effects into independent vectors  $\Delta f_b$  and  $\Delta f_f$ . After these modifications, the functions below can be written, to express the equilibrium of a solid containing cracks:

$$\begin{aligned}
Y_b &= a_b \Delta x + (H_b^{frs} - H_b^{f1s}) \Delta U_{rs}^f + (H_b^{frm} - H_b^{f1n}) \Delta U_{rn}^f + H_b^{f1s} \Delta u_s + H_b^{f1n} \Delta u_n - \\
&- \Delta f_b - (G_b^{frs} + G_b^{f1s}) \Delta f_s^f - (G_b^{frm} + G_b^{f1n}) \Delta f_n^f
\end{aligned} \tag{18}$$

$$\begin{aligned}
Y_f &= a_f \Delta x + (H_f^{frs} - H_f^{f1s}) \Delta U_{rs}^f + (H_f^{frm} - H_f^{f1n}) \Delta U_{rn}^f + H_f^{f1s} \Delta u_s + H_f^{f1n} \Delta u_n - \\
&- \Delta f_f - (G_f^{frs} + G_f^{f1s}) \Delta f_s^f - (G_f^{frm} + G_f^{f1n}) \Delta f_n^f
\end{aligned} \tag{19}$$

In Equations (18) and (19), the matrices  $a_b$  and  $a_f$  contain the coefficients of matrices referred to unknown boundary displacements and tractions. These two last equations can be further modified to emphasize the variables related to cohesive criterion in the non-linear process. These variables are de crack relative displacement and the traction, both in  $n$  direction. Then, the two last equations above can be rewritten as:

$$Y_b = A_b \Delta X + H_b^{f1n} \Delta u_n - \Delta F_b - (G_b^{frm} + G_b^{f1n}) \Delta f_n^f \tag{20}$$

$$Y_f = A_f \Delta X + H_f^{f1n} \Delta u_n - \Delta F_f - (G_f^{frm} + G_f^{f1n}) \Delta f_n^f \tag{21}$$

In these equations, the matrices  $A_b$  and  $A_f$  contain the coefficients of matrices referred to unknown boundary and crack displacements increments ( $\Delta U_b$ ,  $\Delta U_{rs}^f$  and  $\Delta U_{rn}^f$ ), boundary tractions ( $\Delta P_b$ ), and the crack relative displacement at the  $s$  direction ( $\Delta u_s$ ). The known boundary and crack values, this last one only at the  $s$  direction, are taken into account with independent vectors  $\Delta F_b$  and  $\Delta F_f$ . Along the crack surfaces, only the traction components in  $n$  direction are considered in the non-linear process. The traction components in direction  $s$  are neglected according to the cohesive crack model.

The non-linear system of equations given in (20) and (21) can be solved by applying Newton–Raphson’s scheme, for which an iterative process may be required to achieve the equilibrium. By linearizing Equations (20) and (21) and using only the first term of Taylor’s expansion one has:

$$Y_b(\Delta u_{nk}^i) + \frac{\partial Y_b(\Delta X_k^i, \Delta u_{nk}^i, \dots)}{\Delta X_k^i} \delta \Delta X_k^i + \frac{\partial Y_b(\Delta X_k^i, \Delta u_{nk}^i, \dots)}{\Delta u_{nk}^i} \delta \Delta u_{nk}^i = 0 \quad (22)$$

$$Y_f(\Delta u_{nk}^i) + \frac{\partial Y_f(\Delta X_k^i, \Delta u_{nk}^i, \dots)}{\Delta X_k^i} \delta \Delta X_k^i + \frac{\partial Y_f(\Delta X_k^i, \Delta u_{nk}^i, \dots)}{\Delta u_{nk}^i} \delta \Delta u_{nk}^i = 0 \quad (23)$$

The derivate terms in Equations (22) and (23) give the global tangent operator [C]. Thus, carrying out all indicated derivates in these two last equations, the tangent operator for the case of cohesive crack can be obtained. Therefore:

$$[C] = \begin{bmatrix} [A_b] & [H_b^{f1n} - (G_b^{fm} + G_b^{f1n}) \partial \Delta_n^{ff} / \partial \Delta u_n] \\ [A_f] & [H_f^{f1n} - (G_f^{fm} + G_f^{f1n}) \partial \Delta_n^{ff} / \partial \Delta u_n] \end{bmatrix} \quad (24)$$

where the derivate  $\partial \Delta_n^{ff} / \partial \Delta u_n$  is obtained by differentiating properly the cohesive crack laws given in section 2. Thus, the corrections  $\delta \Delta X_k^i$  and  $\delta \Delta u_{nk}^i$  are obtained by solving the linearized system represented by equations (22) and (23):

$$\begin{Bmatrix} \delta \Delta X_k^i \\ \delta \Delta u_{nk}^i \end{Bmatrix} = [C]^{-1} \begin{Bmatrix} Y_b(\Delta u_{nk}^i) \\ Y_f(\Delta u_{nk}^i) \end{Bmatrix} \quad (25)$$

Within a given load increment  $k$  the solution is obtained by cumulating the corrections calculated using Eq.(25)

$$\Delta X_k^{i+1} = \Delta X_k^i + \delta \Delta X_k^i \quad (26)$$

$$\Delta u_{nk}^{i+1} = \Delta u_{nk}^i + \delta \Delta u_{nk}^i \quad (27)$$

After solving the non-linear system of equations in terms of  $\Delta X_k$  and  $\Delta u_{nk}$  all variables have to be updated before applying the next load increment. The tolerance to stop the iterative process within an increment of load is applied on the variation of the crack opening displacement corrections, i.e.,  $u_i - u_{i-1} \leq \textit{tolerance}$ .

## 5 CRACK GROWTH SCHEME

To determine when the cracks grow, the real stress state at the crack tip is compared with an ultimate stress state, given by an adopted criterion. In this paper, the ultimate stresses are calculated using the Rankine's model, which has also been used by (SALEH; ALIABADI, 1995). To achieve accurately the stress state at the tip, a polynomial interpolation process has been adopted to take into account the continuity of the stresses along all points surrounding the crack tip. Several internal points are defined ahead of crack tip, as shown in Figure 3.



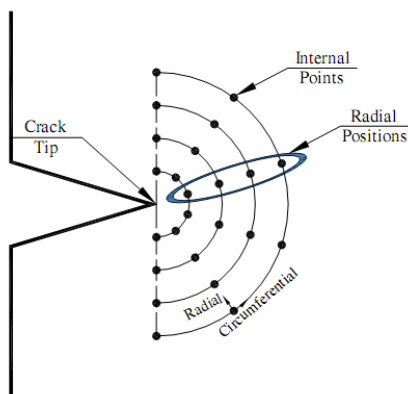


Figure 3 – Distribution of internal points ahead of the crack tip.

The number of semi-circles and the internal points along each one can be chosen according to the desired accuracy. The polynomial degree is chosen according to the number of the semi-circles. For the case illustrated in Figure 3, the interpolation process is performed using third degree polynomial. The interpolation is performed for each radial line and then the stress state at the tip is obtained by extrapolation. The real stress state at the tip is then obtained by averaging the values of all calculated values. The direction of the new crack length increment is given by the direction perpendicular to the maximum circumferential tensile stress. According to this criterion, the crack growth direction is given by:

$$\theta_p = \text{ArcTan} \left( \frac{\tau_{xy}}{\sigma_1 - \sigma_y} \right) \quad (28)$$

where  $\sigma_1$  is the maximum tensile stress value.

The new crack appears when the tensile stress at the crack tip is larger than the ultimate tensile value given by Rankine's criterion. The crack length increment is determined adjusting the size of the element in such a way that the stress state at the new crack tip, calculated by the maximum tensile stress criterion, is equal to the ultimate tensile.

## 6 EXAMPLES

Two examples were chosen to illustrate the efficiency of the proposed numerical model in non-linear crack growth analysis. In the first example, a four point bending concrete structure is analyzed and a four point bending multi-fractured concrete structure is considered in the second example.

### 6.1 Concrete four point bending beam

The four point beam considered in this example is shown in Figure 4. The geometry is given by its length of 675mm, height of 150mm and central notch 75mm deep. The material properties were taken from (GALVEZ et al., 1998), who have performed a laboratory test: tensile strength  $f_t^c = 3.0 \text{ MPa}$ , Young's modulus  $E = 37.000 \text{ MPa}$ , Poisson ratio  $\nu = 0.20$  and fracture energy  $G_f = 69 \text{ N/m}$ .

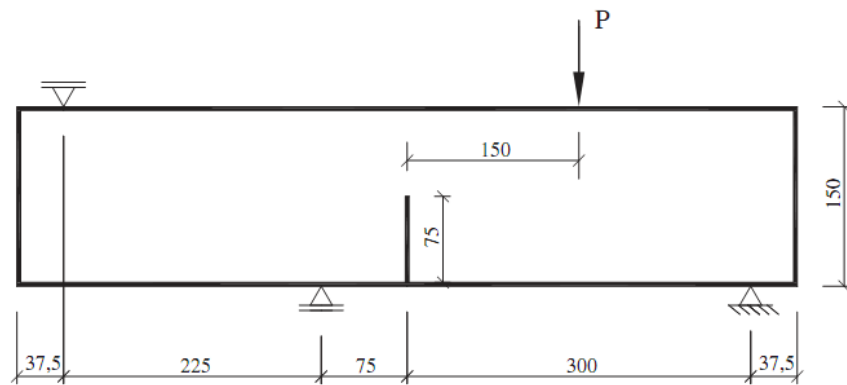


Figure 4 – Four point bending beam. Dimensions in mm.

For the present analysis three cohesive laws were used: linear, bi-linear and exponential. The load was applied for all cases in 24 increments and the adopted tolerance within each increment was  $10^{-5}$ . The two non-linear system solution techniques discussed previously were tested: (a) using constant operator (b) using tangent operator. The results obtained are given in Figure 5 where the tangent operator solutions are identified by the symbol CTO. The other curves are obtained by using the constant operator.

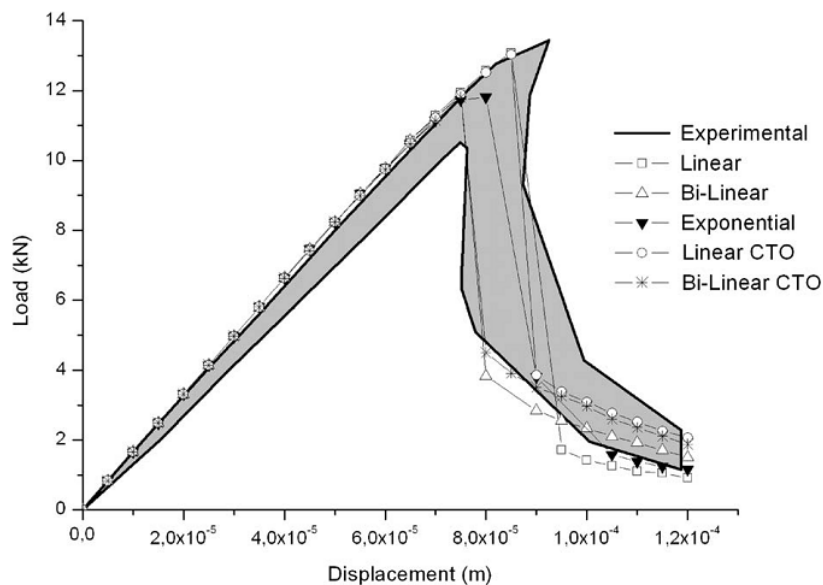


Figure 5 – Load x Displacement curves.

Although all obtained results are in accordance with the experimental values, it seems that the solutions obtained by the tangent operator are more accurate. Moreover, the tangent operator gives always more stable solution requiring a reduced number of iteration at each load increment. It can be seen that the descendent branch obtained by using constant operator is slightly different due to the cumulated errors coming from the large number of post-pick iterations. Figure 6 illustrates the crack growth path during the beam loading, leading to rupture surfaces similar than the ones experimentally obtained. To emphasize the large differences between these two system solution schemes, the iteration numbers to reach the equilibrium at some specific load increments are given in Figure 7. It is important to observe that the constant operator scheme requires a very large number of iterations after pick, and this may lead to less accurate solutions.

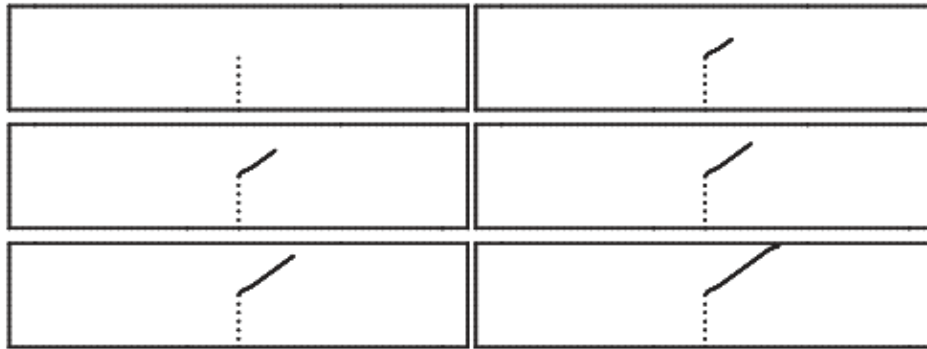


Figure 6 – Crack growth path during the loading process.

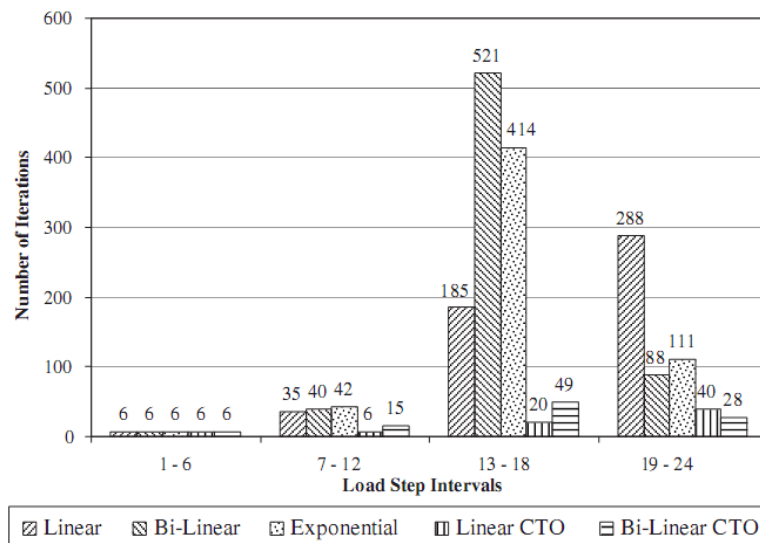


Figure 7 – Required number of iterations for the tested solution techniques.

## 6.2 Multi-fractured concrete structure

In this application, the structure presented in Figure 8 was analyzed. This is a concrete beam subjected to a four bending test. This structure has 1.5 m of length and 0.50 m of high as shown in the same figure. The material properties adopted for this application were: tensile strength  $f_t^c = 3.0\text{ MPa}$ , Young's modulus  $E = 30.000\text{ MPa}$ , Poisson ratio  $\nu = 0.20$  and fracture energy  $G_f = 75\text{ N/m}$ .

In this analysis, eleven cracks were distributed along the lower structural boundary as illustrated in Figure 9. In the same figure is also presented the BEM mesh considered for this analysis. Therefore, the performance of the proposed non-linear formulation in the case of multiple cracks is evaluated. Only two cohesive crack laws were considered in this example: linear and bi-linear. Both cohesive laws were coupled with tangent operator formulation. The load was applied in 25 increments and the adopted tolerance within each increment was  $10^{-5}$ .

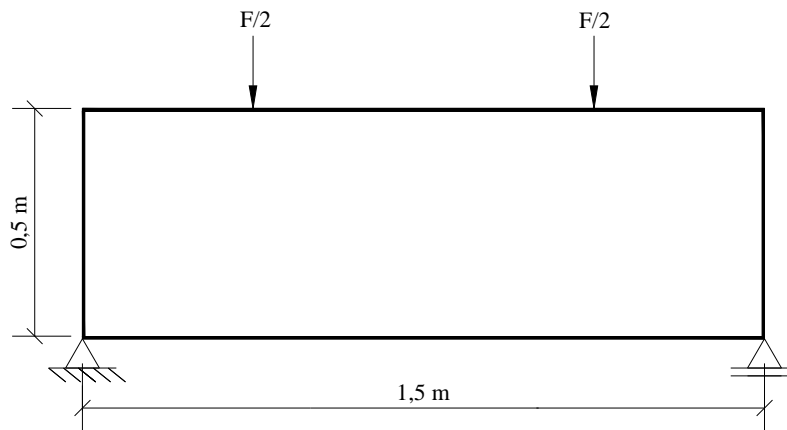


Figure 8 – Dimensions and boundary conditions for the structure.

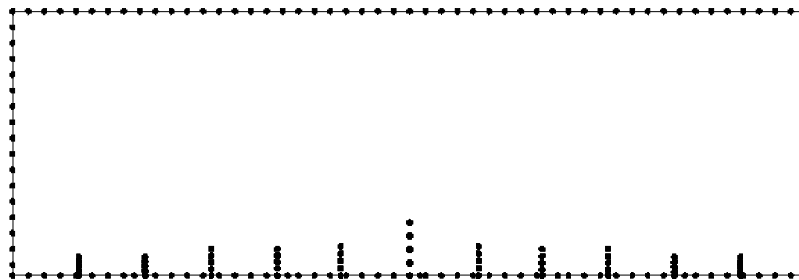


Figure 9 – Crack distribution and BEM mesh adopted.

The load displacement curves obtained in the analysis are shown in Figure 10. According to this figure, it can be observed very similar results to those illustrated in Figure 5, i.e., the structural behaviour achieved by the linear CTO model is more rigid than those obtained by bi-linear CTO model. In both cases the softening branch was verified, after equal results for pre-peak structural behaviour.

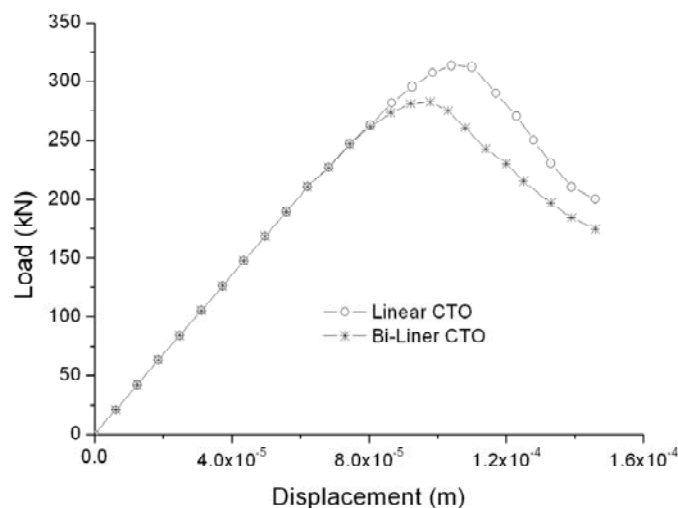


Figure 10 – Load x Displacement curves.

Figure 11 illustrates the crack growth path observed until the last load step applied. According to this figure, it can be observed that only five of eleven cracks have propagated.

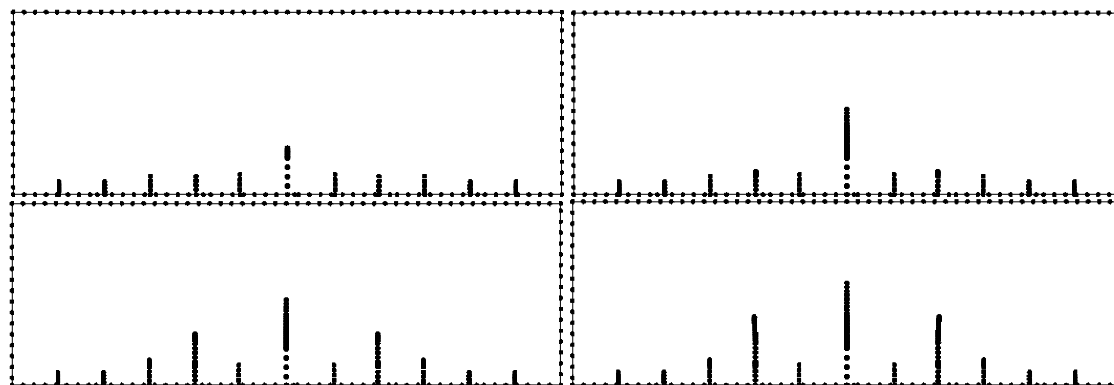


Figure 11 – Crack growth path.

## 7 CONCLUSION

In this paper, the crack growth process in quasi-brittle materials has been studied. This complex structural problem can be modelled by solving a non-linear system of equations which appears due the dependency between crack opening displacement and tractions along the normal direction to the crack lips. To simulate this non-linear structural problem, BEM has shown to be an accurate and efficient alternative. Two non-linear BEM formulations have been developed and implemented in this paper. The first applies a constant operator, where the non-linear problem is solved by keeping constant all relevant matrices and calculating, at each load step, the non-equilibrated vector force. The second approach is developed by using a tangent operator. In this case, the derivate set of the non-linear equations is used and the problem is faster solved. The formulation based on the tangent operator has shown to be more stable and lead to more accurate results in comparison with the classical procedure. The use of tangent operator has shown to be always recommended to analyze crack propagation problems, particularly for the cases where the after pick region is reached.

This kind of formulation may be applied in other non-linear problems in the future. Contact problems and crack growth in reinforced structures, for instance, can be modelled using this efficient approach.

## 8 ACKNOWLEDGEMENTS

Sponsorship of this research project by the São Paulo State Foundation for Research – FAPESP - is greatly appreciated.

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